

SPACE-RUSH

A NEW WAY FORWARD FOR SPACE EXPLORATION & SETTLEMENT, ALIGNED WITH PRESIDENT OBAMA'S NEW SPACE VISION

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Introduction

Since the days of the Apollo program, most of NASA's manned exploration activities beyond LEO have used what could be called a 'mission' based approach. This approach requires that the astronauts take all of the resources required for the mission from Earth, at great expense and difficulty, while providing little (if any) infrastructure for future missions. This approach also relies solely on government funding and direction, with private enterprise engaged only as subcontractors.

In February 2010, President Obama made radical changes to this historical approach to manned space exploration, including canceling the Constellation program's Ares 1 and 5 rockets, the Orion capsule, and the Altair Lander.

Some of the highlights of the President's report include:

- ☐ Extending the life of the ISS until at least 2020.
- ☐ Relying on private, commercial space companies for crew access to low-Earth orbit.
- ☐ A new focus on technology development for exploration and commercial space, and
- ☐ Adoption of a 'flexible path' architecture for human spaceflight, with multiple destinations including near Earth asteroids, Lagrange Points, and eventually the moons of Mars and Mars itself.

There has been considerable controversy concerning this significant change in direction and approach. Many have suggested that this means that the US is giving up on manned spaceflight and returning to the Moon, but we strongly disagree. The Augustine Committee report clearly indicates that the NASA Constellation program was "*on an unsustainable trajectory*" and would not succeed in returning astronauts to the surface of the Moon until late in the next decade, perhaps by 2028, if then.

The new approach put forward by the President is much more than just technology development. It is a strategy for embracing the challenge of sustainable exploration and settlement, ones that goes beyond 'flags and footprints.' It's the difference between the Lewis & Clark expedition and building the transcontinental railroad.

As we have seen from other periods of human expansion, including the settling of the American West, the door to expansion and development opens when the government establishes a minimum level of infrastructure, which in the case of the West was railroads and forts, after which entrepreneurs and settlers quickly followed. This then enabled a significant reduction in the resources required by the hardy pioneers who followed, and reduced the difficulties they faced.

When deliberate actions are undertaken to engage private enterprise (e.g., the Pacific Railway Act and the Homestead Act of 1862), entrepreneurs then flock in and find ways of making productive use of the new resources that the frontier offers. These entrepreneurs also bring the benefits of the frontier back to the settled communities, providing proof of the value of the new terrain, and creating additional excitement. Taken together, establishment of infrastructure and engaging private enterprise set the stage for very effective and rapid expansion and development. For example during the California gold rush between 1848 and 1850, the population of San Francisco grew from 1,000 to 35,000 and the cost of a housing lot skyrocketed from \$16 to \$45,000.

This route to rapid expansion and development is as true today as ever, as we see with the recent opening of an important technological frontier, the Internet. It was the government, specifically the Department of Defense's Advanced Research Projects Agency (ARPA) that put in place the first high-speed data lines between powerful computers in the US to support government and university researchers in the late 1960s and 1970s. Had this high-speed data transfer capability remained the sole province of the government, few of us outside of select government and university scientists would ever have known of its existence, or its utility.

But the door to this high-speed network was opened to resourceful entrepreneurs in the 1980s, and the Internet was born. Few could have anticipated the enormous economic and public benefit that accompanied the advent of e-commerce and search engine technologies. Today, the internet advertising industry alone adds \$300B to the US economy, amounting to 2.1% of US GDP, and this from an industry that did not exist 15 years ago. As with many others, we believe that the economic benefits to be had from the commercial development of space could be significantly larger.

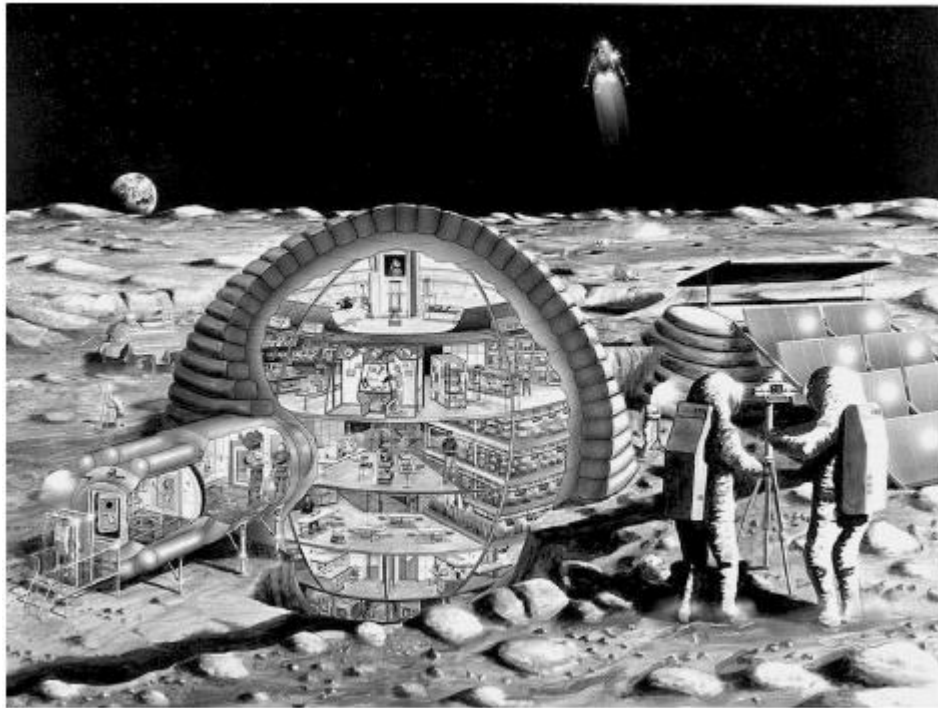


Figure 1

NASA illustration showing commerce and habitation on the moon.

A New Way Forward for Exploration

In laying out a path for space exploration for this new millennium, it is essential to learn from the past and use an effective strategy to enable rapid

expansion and economic development. Such a strategy has critical tasks for both the government and private enterprise.

The government must enable the demonstration of critical capabilities, the gathering of critical knowledge, and the development of key infrastructure. Once these initial activities and developments are accomplished, the government must not only allow but actively encourage private sector engagement and investment to enable timely, cost effective, and productive uses of the new space frontier that show a direct benefit to the people here on Earth.

This may include new services such as space tourism and entertainment, new knowledge such as fundamental new understandings of biology and materials, new products including new medicines, vaccines, and materials, and new resources including space based power. This will enable the general public to understand and appreciate the economic and social value of space, which in turn help to maintain NASA's relevance. We call this infrastructure based, private enterprise engaged strategy for space exploration 'Space-Rush.'

Five categories of infrastructure can be identified that would greatly reduce the resources required for space commercialization, and hence the difficulties that space entrepreneurs will encounter:

1. Optimized Transportation
2. Power and Communications
3. Fuel and Water
4. In-situ Resource Utilization
5. Crew Accommodations

Each of these categories is examined below, including a strategy for establishing it, and options for engaging private enterprise for the initial stages, optimization, and broad utilization.

With the capabilities from these infrastructure elements in place and private enterprise effectively engaged, the potential for important new products and benefits for Earth would be greatly increased.

Optimized Transportation

NASA has tried repeatedly to develop a 'next generation' replacement for the Space Shuttle over the last two decades, and the story reads like a litany of missed opportunities. The National Aerospace Plane (X-30), VentureStar (X-33), X-34, National Launch System, the Space Launch Initiative, and most recently Constellation and the Ares 1 and 5 all started and failed. Billions of dollars have been spent, and millions of man-hours expended, with very little to show for it. Tragically, few of these vehicles got even an inch off the ground.

Albert Einstein noted that a sure sign of insanity was when people keep doing the same things over and over but expect to get a different result, so we must ask what we can learn from these experiences that will help us get it right this time? It's time for us to take this to heart and try some new approaches in space transportation development, with the goal to develop an affordable and reliable orbital space transportation system, one that actually gets to space.

We need to think beyond getting to Low Earth Orbit (LEO), and open space for public access and commercial development, including access to the Moon and beyond. This means considering four different transportation requirements:

- ☐ Earth to Orbit (ETO) and return
- ☐ Orbit to Orbit (OTO), including LEO to Geo-synchronous (GEO),
- ☐ Earth orbit to lunar and Mars orbit, and finally
- ☐ Orbit to Extraterrestrial Surfaces (OTES), including the lunar surface, asteroids, Phobos and Demos, and eventually Mars.

While NASA has been focused almost exclusively on rocketry and ETO transportation, in addition to rockets there are many intriguing propulsion options that have not been explored to any significant degree, including laser propulsion, in which ground or space based lasers are used to propel a vehicle to orbit, and spinning tethers that could be used to rendezvous with reusable suborbital rockets. Both of these concepts could reduce the size, complexity, and cost of launch vehicles.

One of the key goals for optimized space transportation is to develop fully reusable vehicles that will significantly reduce the cost of space travel.

Technical performance is not the key to reusable vehicles. The Space Shuttle is an example of a vehicle that, while refurbishable, is expensive to operate and maintain. Even with the significant reductions in maintenance staff between 1995 and 1999, the Shuttle still requires 1800 workers to keep it operational.

In the design phase for next generation vehicles, emphasis must be placed on efficient maintainability and operability so that space systems operate more like commercial airlines, and less like experimental aircraft.

Several suborbital vehicles are now under construction with the goal to travel to 100 km or more on a routine basis; some are even planning multiple flights per day with a ground crew of fewer than 10. These companies include XCOR Aerospace, Armadillo Aerospace, Masten Space Systems, Blue Origin, and Virgin Galactic.

Fuel depots are another option that has recently received a good deal of attention because of the Augustine Committee review of US Human Space Flight Program. The concept is straightforward: a fuel depot is a filling station in space, so instead of taking all the fuel you need for your

space mission, you take only enough to get to the depot, where you refuel just as you refuel your car on road trips. This allows the fuel for the depot to be launched on less expensive, reusable launchers, and as secondary payloads on missions that have excess payload margin.

Fuel depot technology has advanced significantly, and we now have the technology to store liquid oxygen and even liquid hydrogen for extended periods of time in space with very low boil-off, a key requirement for storing fuels on orbit for long duration.

Using a fuel depot allows the launch vehicles needed for space missions to be significantly smaller, potentially eliminating or at least reducing the need for a heavy lift vehicle.

There is also an interesting synergy between the development of fuel depots and reusable launch vehicles (RLVs). The key to keeping costs down on reusable launch vehicles is frequency of flights. If fuel depots are resupplied by commercial RLVs then this will help provide the demand needed to close their business case.

*(Editor's note: A somewhat less optimistic view of the fuel depot concept is presented in Chapter 13, *Prospects for In-Space Re-Fueling*.)*

OTO transfer is conceptually much simpler than ETO travel. OTO can be subdivided into a couple of categories. The first of these is LEO to GEO transport. The idea of a reusable space tug to provide LEO to GEO transportation has been studied for decades. The Orbital Transfer Vehicle (OTV) and its several derivatives were studied by NASA, Lockheed, and Boeing in the 1970s and 80s, but were never developed for budget and programmatic rather than technical reasons. OTVs could also make use of fuel depots, and they could also utilize aerobraking to slow OTVs down by using the Earth's atmosphere to slow the OTV down when it returns from GEO to LEO.

The original NASA concept was to house OTVs at a space station. Payloads would be launched from Earth on an ETO vehicle that would dock at the Station. There, the payload would be removed and attached to the OTV. The OTV would then take the payload to GEO orbit, where it would be released. The OTV would use its aerobrake to return to the station.

A velocity change of about 4 km/sec is required to get from LEO to GEO, as well as a required plane change to the angle of orbit. Coincidentally, this is the same velocity change that is needed to get to Low Lunar Orbit (LLO) from LEO (4.04 km/sec) (see figure 2). Therefore the same type of OTV that is used for Earth orbit and GEO could also be used to transfer payloads to and from lunar orbit.

In addition to chemical propulsion, advanced propulsion technologies offer the potential of improved performance that could open up the solar system for commerce and settlement. Ion propulsion

technology has already been demonstrated in space with the Deep Space 1 and Dawn missions to the Moon, Mars and the asteroid belt. Although the thrust is low, often a fraction of a pound, the propulsion efficiency (Isp) is very high, and as it runs continuously, after weeks or months very high velocities can be achieved. This is fine for cargo, but not useful for transporting people.

There is also a particular problem when leaving LEO due to the extended time the vehicle spends in the Van Allen radiation belt.

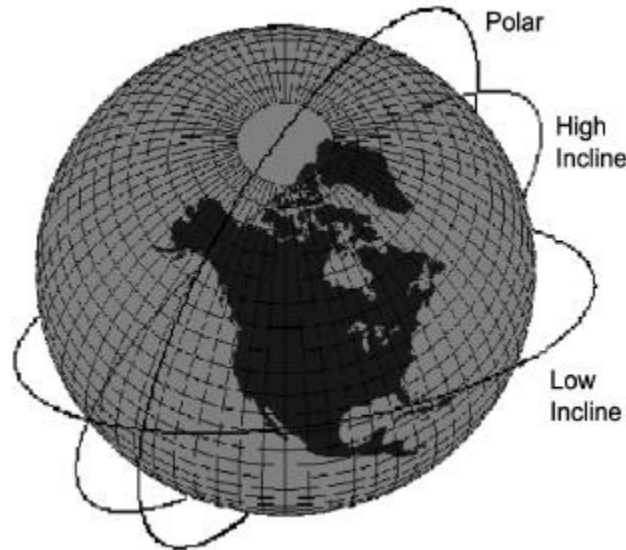


Figure 2
General categories of orbital inclination planes.

What is needed is another option that could be used to propel the OTV from LEO to GEO or LLO.

To open up the solar system, what is needed is a high thrust system that also has high Isp, and there are a few possibilities that could be quite attractive. Former astronaut Dr. Franklin Chang Diaz and his company Ad Astra have developed the VASIMR (Variable Specific Impulse Magnetoplasma Rocket) concept, which will produce both high thrust and high specific impulse. A VASIMR prototype has been tested in the laboratory up to 200-kilowatts, and worked as expected.

Ad Astra has signed a Space Act Agreement with NASA to fly the VASIMR engine on the International Space Station. A 200MW VASIMR engine could propel a manned mission to Mars in less than 40 days, compared to the 270 days it would take using chemical propulsion. We will discuss how to achieve these power levels in the next section.

The last transportation segment to consider is from OTES and back. A good example is LLO to the lunar surface. Since there is no air on the moon, a hypergolic propulsion system similar to what was used on

Surveyor and Apollo could be used, and there are interesting options to consider. One possibility is to harvest rocket fuel from the lunar soil in a process called in-situ resource utilization (ISRU).

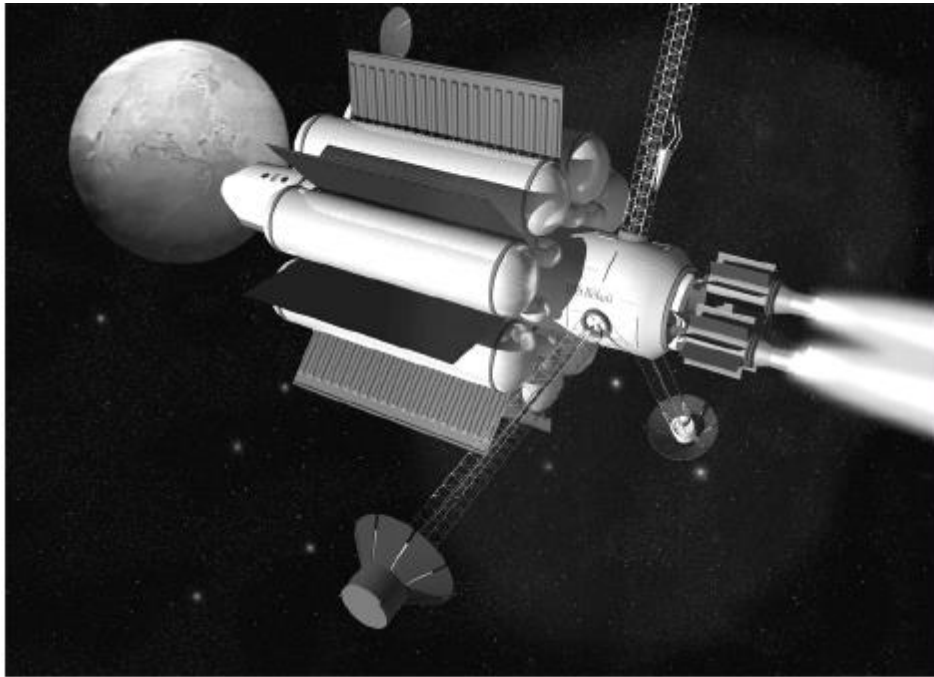


Figure 3

Human Mission to Mars with a Nuclear Powered VASIMR engine.
Illustration courtesy of Ad Astra.

As the Moon is ~45% oxygen by mass, rather than transporting oxygen from Earth to the lunar surface at great expense, extracting the oxygen required for the return trip from the lunar soil is an attractive option. Twenty-seven different possible processes have been identified for performing such an extraction.

A great benefit of this approach is that by-products of oxygen extraction include iron, silicon and titanium, all of which are of course valuable materials.

Another, more technologically advanced option would be a Lunar Elevator that would go from the Earth/Moon L1 point to the Lunar surface, a distance of 56,000 km. While building a space elevator on Earth is beyond the current capabilities of materials technology, this is not the case on the Moon with its lower gravity and lack of atmosphere. A cable would be lowered from L1 to the Lunar surface and a climber vehicle attached to the cable. Using beamed power (laser or microwave), the climber would ascend the cable, completing the journey in a matter of a few days at very low cost. The process would then reverse, and cargo and people would descend from the L1 terminus to the Lunar surface.

Technologies including mass drivers and rotating tethers will be the subjects of further R&D for their possible application to the Moon, where they may be particularly well suited.

A mass driver is a long electro-magnetic accelerator with a series of coils through which a conducting payload canister is accelerated. It is kind of like a linear motor. A rotating tether is like a two rocks rotating on a common string. If the string is cut at the right time they will go flying off in opposite directions. In orbit, one would go to a higher orbit and one could be re-entered into the atmosphere.

Devices such as these may ultimately make the mining of such high value materials such as Helium-3 (which we will discuss in the next section) economically viable.

Power and Communications

Power and communications are the next essential infrastructure elements. One concept that has particular potential for both government initiation and private enterprise is in-space power beaming. Using current US government capabilities in high-power lasers, and US industry's large satellite design capabilities, several hundred-kilowatt space power stations could be designed and constructed incorporating new high efficiency photovoltaic solar cells and other relevant technologies. These space power satellites could then beam power to locations of space activities, such as the lunar surface or Lagrange points. These space power stations could also serve as communications hubs, transferring broadband data to and from points of interest.

A government directed demonstration program could accomplish the necessary initial infrastructure for these space power and communication stations. One very useful activity would be to construct a demonstration space power and communication station to support NASA's lunar surface missions. With successful demonstration of the necessary technologies and systems, construction and deployment of operational power and communication power stations could be handed off to industry.

Borrowing from the NASA Commercial Re-Supply (CRS) program for the ISS, an agreement could be offered through which the government agrees to purchase minimum levels of power and data communication from the stations.

To go beyond the moon will require large amounts of power for such systems as the VASMIR. Power levels from 100MW to 10GW would be required to truly open up the solar system for exploration, commerce and settlement.

Power systems for space propulsion differ in two key ways from terrestrial, stationary counterparts. The first is that output power is not the only consideration, as energy density, measured as kilowatts per kilogram,

is also a vital factor. The other key difference is that getting rid of waste heat is much more difficult in space than on Earth, so energy conversion efficiency becomes important. In space both conduction and convection are not viable for getting rid of waste heat, and this leaves only radiation, but the radiators required to dispel waste heat can add a very severe mass penalty to the design of a vehicle, and also severely handicap its performance.

In the past, the primary mechanisms for providing power to space systems has been solar energy and nuclear fission. Solar energy is typically collected by solar cells that convert incoming photons into electricity. This works well for power levels up to 10-100 kilowatts, but for higher power levels the arrays become very large. But as one travels outward from Earth's orbit toward Mars, the energy flux (energy/unit area) from the sun decreases, requiring an increase in the size of the solar array required to produce the same amount of power.

Nuclear fission is the other approach that has been used. Radioisotope Thermoelectric Generators (RTGs) have been used on all of the unmanned outer planet probes, including Pioneer, Voyager, Galileo, and Ulysses. These devices generate heat from the nuclear decay from the fissionable material, typically plutonium, and then convert this heat to electricity using thermocouples. The energy conversion efficiency of these devices is very low (3-7%), but they are very simple and reliable. These systems work well for low power levels less than one KW, but do not scale well.

However, there are other technologies under development that can help meet these challenges. Two options discussed here are Nuclear (fission) Thermal Rocket (NTR), and A-neutronic Fusion Rockets (AFR). The concept for a NTR is to use the fission reactions in the reactor core as a heat source, run fluid through the core to heat it up and then expand this hot gas/plasma out the back as fast as you can. From the late 1950s through the early 1970s the US spent \$1.4B on solid-core nuclear rocket R&D, and more than 20 NTR reactors were designed, built, and tested at the Nevada Nuclear Test Site at Jackass Flats Nevada.

These engines achieved exhaust temperatures of 2,350-2,550 K using graphite fuel and an Isp of 825-850 seconds with burn durations from 62 minutes to over 4 hours, and an engine thrust to weight of ~3. The technology of these engines was relatively mature in 1970s, and some work has continued by companies such as Aerojet as recently as 2002, so it may not be a significantly difficult feat to develop such an engine for space applications. The public relations difficulty, however, concerning launching nuclear material is another matter altogether.

The second option is the AFR, which differs from the conventional approach to achieving fusion that the US has pursued for the last half century. This conventional approach attempts to burn a mixture of tritium

(a radioactive gas) and Deuterium (DT), both of which are isotopes of hydrogen. This fuel combination is the easiest one in which to induce fusion (13.6 keV required), but even if DT fusion is achieved, it presents several drawbacks, including the fact that 80% of the energy released in this reaction comes out as energetic (14 MeV) neutrons and only 20% comes out as charged particles. These high-energy neutrons cause many problems, including inducing radioactivation (the neutron flux changes the atomic structure of the surrounding structure) that causes the material to become radioactive as well as degrading material strength. The only way these neutrons can be turned into useful energy is to thermalize them (using a large blanket of liquid lithium for instance) and then running this hot fluid through a steam cycle to produce electricity. Due to materials limitations, the Carnot efficiency of such processes is very low (<20%), and it produces a very large waste heat problem for any space application.

Fortunately, there has been significant recent progress in a different approach, aneutronic fusion. Aneutronic fusion differs from conventional fusion reactions in that neither of the fuel elements is radioactive, and the resulting fusion products are charged particles. Two aneutronic reactions are of particular interest for space applications, DT and He3 (D-He3), and Protium (ionized hydrogen) and Boron (P-B11). D-He3 is easier to burn – 58 keV compared to 123 keV for P-B11. Since this is a more difficult technical challenge, new approaches need to be tried to reach these high energies. However, the US Department of Energy (DOE) has repeatedly refused to provide significant support to these advanced concepts despite numerous calls by Congress to do so. Some efforts are nevertheless under way, but most are underfunded, which of course limits their progress.

Another interesting aneutronic fusion concept is inertial electrodynamic fusion (IEF). P. T. Farnsworth and Robert Hirsch developed the basic concept for IEF in the 1960s as a spherical accelerator. Electrostatic potentials are used to accelerate the particles to velocities where their momentum can overcome the coulomb barrier and fusion can occur. Materials limitations prevented the Farnsworth/Hirsch device from producing net power, until in the 1980s when Dr. Robert Bussard modified the Farnsworth/Hirsch device and replaced their electron grid with a magnetically-insulated ‘magrid.’ Over the next two decades with funding from the Navy (although none from DOE) Bussard’s company EMC2 was able to demonstrate many of the fusion requirements for a practical fusion device, including producing 109 fusion reactions/sec at very low voltage (10 kV). Unfortunately, Dr. Bussard passed away in 2007, but his work is being carried on by Dr. Richard Nebel.

In 2009 the Navy awarded EMC2 a \$12M contract that, if all the options were exercised, would demonstrate PB11 fusion in 2012. Dr. Nebel predicts that the next power producing system could be demonstrated by 2020, a system that could be capable of producing

hundreds of megawatts to gigawatts of electrical power, with no radioactivity, and at conversion efficiencies as high as 95%.

Shortly before his death Dr. Bussard made a presentation at the International Space Development Conference on space applications using IEF technology, and he predicted that IEF could power a colony on Mars capable of housing 1200 people with 50 tons of supplies each, for under \$20B.

Several other a-neutronic fusion concepts have received public or private funding, including efforts by Tri Alpha and FRC machine, Lawrenceville Plasma Physics and their Dense Plasma Focus, Magnetized Target Fusion at Los Alamos National Lab, as well as a Sandia Labs and Prometheus II Ltd. PLASMAK \square device. Each of these devices has unique advantages and challenges, but experimental work has been done and the results have been encouraging enough to continue development. Most if not all of these concepts could provide the energy levels and propulsion performance that could open up the solar system for commerce and settlement.

Fuel and Water

As previously noted, space fuel depots could constitute an important new space infrastructure. A primary function will be to store liquid oxygen and hydrogen, as well as other expendables, including water. Ample supplies of hydrogen and oxygen enable you to produce water easily in standard fuel cells that also then provide electrical power. A space depot could therefore be an important source of fuel, water and even food grown in space based greenhouses, which would support a wide range of activities of interest to both government and commercial firms. With access to the lunar surface or suitable asteroids, in-situ resource utilization (ISRU) techniques could be employed to provide additional sources for fuel, water, and even building materials.

Government demonstrations of space depots should be the first step, and when the techniques and technologies are proven the government could then turn over construction and operational of space fuel depots to industry. As with power and communication contracts, minimum government purchase agreements for fuel, water, and even food, could provide a guaranteed market to ensure and stimulate commercial adoption of these facilities.

Recent discoveries by NASA's Moon Mineralogy Mapper (MMM) on board India's Chandrayaan-1 Lunar Orbiter, and by the US Lunar Reconnaissance Orbiter (LRO) and Lunar Crater Observation and Sensing Satellite (LCROSS) probe, reveal data that suggest that there is at least 600 million tons of water contained in ice sheets 1 to 3 meters thick on the moon's north pole. This invaluable resource could be used to support a

variety of purposes, including providing oxygen to breathe, water to drink and grow crops, and hydrogen and oxygen for propulsion and many other industrial purposes, including re-supply of orbiting fuel depots.

In-Situ Resource Utilization (ISRU)

Living off the land has historically been the key to opening up new frontiers. When American settlers moved west of the Mississippi they came upon the Great Plains with not a tree in sight. To create shelter they could have hauled wood to build traditional houses, but that would have been prohibitively expensive. Instead they used local materials such as sod and adobe to build a new kind of structure that could keep cool in the summer and warm in the winter. Similarly, space settlers will need to learn to live off the resources that they find wherever they go.

While the moon is mostly oxygen, it also contains other valuable materials including iron, silicon, aluminum, calcium, magnesium, sodium, and titanium. The data from LCROSS from the Oct. 9, 2009 impact shows definitively that there is water ice in the permanently sheltered craters on the poles of the moon as well as other economically valuable materials.

The abundance of solar radiation on the moon could be readily applied to breaking down the compounds into a useful form. For example, as noted above, 27 different processes have been identified for extracting oxygen from the lunar soil, and a great side benefit of many of them is that the by products are iron, silicon, and titanium, all of which are very valuable in their own right.

The asteroids are also sources of mineral wealth. While most of the known asteroids are located far from Earth in orbit between Mars and Jupiter, there is a class of asteroids called Apollo objects which have orbits that come very close to and in some cases cross the orbit of the Earth. These objects are a potential source of raw materials, but they also pose a threat because a collision with even a small object (100 m) could be catastrophic, suggesting that it is important to learn more about these objects for profit and also for protection.

Hence, Congress has tasked NASA with “*detecting, tracking, cataloging and characterizing near-Earth asteroids and comets in order to provide warning and mitigation of the potential hazard of such near-Earth objects (NEOs) to the Earth.*” In response, NASA established a program to identify and track NEOs greater than 140 meters in diameter.

As of August 2009, 6,244 such objects had been cataloged. Asteroids make up the majority of these objects, and there are two major types: 1) metallic and 2) carbonaceous chondrites. A typical metallic asteroid is composed of iron and nickel, both valuable elements. A single 1 km wide metallic asteroid could provide the earth with enough iron and nickel to meet the current world demand for 2 - 3 years. Carbonaceous

chondrites, on the other hand are made of silicates, oxides and sulfides, but more importantly a significant portion of them contain water (from 3-22%) and other volatiles.

From an energy point of view, many of these NEOs require even less delta V to reach them than is required to land on the moon, which would conceivably make it possible to mine these objects for valuable materials elements, and then return them to cis-lunar space for use and economic benefit.

Crew Accommodations

Expandable space habitation modules can be purchased today from commercial providers such as Bigelow Aerospace. The Bigelow expandable space habitat is a success story that should be noted and copied. The design of this deployable habitat was adapted from a NASA advanced technology program called Transit Habitat, run by Johnson Space Center in the 1990s, with the intention to design an interplanetary vehicle to transfer humans to Mars. The Transhab concept that emerged from this project was intended as a replacement for the already existing International Space Station crew Habitation Module. But while the ISS habitation module is a rigid structure, inflatable modules can be launched in a compact form. When fully inflated, Transhab would expand to 8.2 meters in diameter (compared to the 4.4 meter diameter of the Columbus ISS Module).

Controversy arose during Transhab development due to delays and increased costs of the ISS program, and the National Space Society issued a policy statement recommending that NASA cease development of Transhab. Finally in 2000, House Resolution 1654 was signed into law banning NASA from conducting further research and development of Transhab, but an option to lease an inflatable habitat module from private industry was included in the bill.

Since that time, Bigelow Aerospace has purchased the rights to the patents developed by NASA, and is pursuing a similar scheme for a private space station design. The company has launched the Genesis I and Genesis II pathfinder spacecraft, with plans for additional experimental craft culminating in their BA 330 production model. Bigelow plans to launch the first series of expandable modules to orbit in 2014, and to welcome the first inhabitants in 2015. By 2020 he could accommodate as many as 24 people in orbit at one time.

The government may be interested in purchasing expandable habitats for space, and even for lunar sorties and outposts it would seem to be a very effective approach to crew accommodation. Developing and using ISRU to help provide oxygen, fuel, building materials and even food would be a natural extension of this concept. Continued government supported for research and development on advanced technologies and concepts, for

crew accommodations and life support systems would also be a smart investment. As history has shown many times, government developed concepts, like Transhab, that are proven and then transferred to industry, can lead to rapid and significant benefits for the government, industry, and the general public.

Summary

History tells us that infrastructure-based exploration led by the government, with active engagement of industry, provides the best opportunity for economic expansion in the space frontier. As the expansion of the railroads and Internet shows, this strategy offers the best means of providing the necessary foundation for development while engaging the entrepreneurial spirit of the private sector for economic and social benefit.

The ‘Space-Rush’ strategy described here recommends focusing on four critical infrastructure elements:

1. Optimized Transportation,
2. Power,
3. Fuel, Water and Materials, and
4. Crew Accommodations.

Using government-funded demonstrations, followed by guaranteed purchase agreements to assure minimum demand while engaging private entities provides a proven path for success.

Important activities that could be pursued immediately include space fuel depot development, in-situ resource utilization experiments, space power beaming demonstrations, and expandable crew accommodation purchases. Re-establishment of long-range government funded technology programs should also be pursued to assure that new and improved technologies are continually under development.

With ‘Space-Rush’ we will finally begin to open space to rapid and beneficial development for the government, industry, and the general public.

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Bruce Pittman



Bruce Pittman is the Director of Flight Projects and Chief System Engineer at the NASA Emerging Commercial Space Office at the Ames Research Center, where he supports programs ranging from suborbital human-tended research, orbital applications and research, low cost, responsive access to space, and lunar commercialization. He has been involved in high technology product development, project management and system engineering for over 30 years.

He started his career at NASA Ames for 11 years on projects including Pioneer Venus, the Infrared Astronomy Satellite (IRAS), the Cryogenic Grating Spectrometer on the Kuiper Airborne Observatory, the Space Station Freedom Technology Advocacy Group, and several advanced studies programs. He has also worked with NASA as a contractor on projects including the NASA Emerging Commercial Space Office (2005-present), Commercial Orbital Transportation Services (2006-2007), International Space Station Commercialization (2005-2006), High Speed Civil Transport (1997-1998), Program and Project Management Initiative (1988-1993), the Space Exploration Initiative (1989-1991) and Space Shuttle Processing (1987-1988).

Mr. Pittman has also been a founder and member of the startup team of early stage growth companies including SpaceHab, Kistler Aerospace, New Focus, Product Factory, Prometheus II Ltd., and Industrial Sound and Motion.

Mr. Pittman has a BS in Mechanical Engineering from U. C. Davis and a MS in Engineering Management from Santa Clara University. Mr. Pittman is an Associate Fellow of the American Institute of Aeronautics and Astronautics (AIAA) a member of the AIAA Commercial Space Group and founder and first chairman of the System Engineering Technical Committee. He is also a member of the organizing committee for the Space Investment Summit series, a member of the Aerospace Technology Working Group (ATWG), and President of the Silicon Valley Space Club. He has authored or co-authored more than 3 dozen papers on a technical, management and business topics in aerospace and high technology. In addition to his technical work Mr. Pittman is also a member of the adjunct faculty in the Graduate Engineering School at Santa Clara University.

For his technical work Mr. Pittman has been awarded 2 NASA Special Achievement Awards, four NASA Group Achievement Awards, and the AIAA Distinguished Leadership Award.

Dr. Daniel J. Rasky



Dr. Daniel J. Rasky is the Director for the Emerging Commercial Space Office at NASA Ames, and also a Senior Scientist with NASA. He is a Co-Founder and Director for the Space Portal whose mission is to “Be a friendly front door for emerging and non-traditional space companies.” He recently completed a one-year Interagency Personnel Assignment (IPA) with the Space Grant Education and Enterprise Institute (SGEEI), where he served as a Senior Research Fellow supporting a number of emerging space companies and other organizations. This included provided expert consulting to SpaceX on the design and development of the heatshield for their Dragon capsule. SpaceX has chosen to use the PICA heatshield material, invented by Dr. Rasky and associates at NASA Ames, for Dragon.

Dr. Rasky is an internationally recognized expert on advanced entry systems and thermal protection materials, with 25 years of experience in advanced entry systems and materials for NASA (20 years) and the US Air Force (5 years). Dr. Rasky has made significant contributions to flight hardware on seven NASA missions, including co-inventing the PICA heatshield material that enabled the NASA Stardust comet sample return mission, and is the primary heatshield for the Mars Science Laboratory (MSL) lander mission.

Dr. Rasky is the recipient of the NASA Inventor of the Year Award (the first ever for NASA Ames), the Senior Professional Meritorious Presidential Rank Award, the NASA Exceptional Achievement Award, the NASA Exceptional Service Medal, twelve NASA Group Awards, and eight Space Act Awards. He has 6 patents, 64 publications, is an Associate Fellow of the AIAA and Senior Member of the ASME.

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